

### **Measuring water level in rivers and lakes from lightweight Unmanned Aerial Vehicles**

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 The assessment of hydrologic dynamics in rivers, lakes, reservoirs and wetlands requires measurements of water level, its temporal and spatial derivatives, and the extent and dynamics of open water surfaces. Motivated by the declining number of ground-based measurement stations, research efforts have been devoted to the retrieval of these hydraulic properties from spaceborne platforms in the past few decades. However, due to coarse spatial and temporal resolutions, spaceborne missions have several limitations when assessing the water level of terrestrial surface water bodies and determining complex water dynamics. Unmanned Aerial Vehicles (UAVs) can fill the gap between spaceborne and ground-based observations, and provide high spatial resolution and dense temporal coverage data, in quick turn-around time, using flexible payload design. This study focused on categorizing and testing sensors, which comply with the weight constraint of small UAVs (around 1.5 kg), capable of measuring the range to water surface. Subtracting the measured range from the vertical position retrieved by the onboard Global Navigation Satellite System (GNSS) receiver, we can determine the water level (orthometric height). Three different ranging payloads, which consisted of a radar, a sonar and an in-house developed camera-based laser distance sensor (CLDS), have been evaluated in terms of accuracy, precision, maximum ranging distance and beam divergence. After numerous flights, the relative accuracy of the overall system was estimated. A ranging accuracy better than 0.5 % of the range and a maximum ranging distance of 60 m were achieved with the radar. The CLDS showed the lowest beam divergence, which is required to avoid contamination of the signal from interfering surroundings for narrow fields of view. With the GNSS system delivering a relative vertical accuracy better than 3-5 cm, water level can be retrieved with an overall accuracy better than 5-7 cm.

Keywords: UAV; water level; radar; sonar; laser; GPS;

#### **1. Introduction**

 Extreme hydro-climatic events such as droughts, floods and heavy precipitation have increased the awareness that knowledge of spatial and temporal variation of open water surfaces is important (Alsdorf et al., 2007). In order to achieve a better quantitative understanding of hydrologic processes and to increase sharpness and reliability of hydrologic predictions, observations of hydrological variables, such as surface water area, water level (h), its slope (∂h/∂x) and its temporal change (∂h/∂t) are required. However, ground-based measurements of terrestrial water bodies are limited to networks of measuring stations. In-situ stations provide point observations that are often spaced too far apart to capture spatial patterns. Often, in-situ observation technology fails during extreme events. Furthermore, globally, the availability of in- situ hydrologic observation stations has been declining in the recent past (Lawford et al., 2013). Hence, remote sensing datasets have become increasingly popular in hydrology. Remote sensing techniques are presently unable to observe river discharge directly, however spatial and temporal variation of water level has been routinely observed using spaceborne or airborne platforms. Although most satellite altimetry missions were not designed primarily for monitoring continental waters, water levels of continental water surfaces retrieved by Seasat, TOPEX/Poseidon, Jason-1 and 2, GFO, ERS 1 and 2, ENVISAT have a measurement accuracy that is well understood and generally on the order of a few tens of centimeters (Calmant et al., 2008). This accuracy can be improved for larger lakes and rivers by averaging over large water surfaces (Birkett, 1998; Birkett et al., 2002; Frappart et al., 2006). The satellite CryoSat-2 carries a Synthetic Aperture Interferometric Radar Altimeter (SIRAL) which is a new generation radar altimeter (Wingham et al., 2006) with a spatial resolution of around 300 m (Villadsen et al.,

 2015). When operating in SARIn mode, a correction of the cross-track slope can be performed and waveform analysis allows separation between water and surrounding topography (Kleinherenbrink et al., 2014) resulting in an accuracy of the retrieved water level of just a few decimeters (Kleinherenbrink et al., 2015). Spaceborne LIDARs such as the Geoscience Laser Altimeter System (GLAS) have been shown to provide water level measurements with higher accuracy than radar altimeters such as TOPEX/Poseidon (Zhang and Xie, 2010). Still, GLAS has a ground footprint that is around 65 m (Schutz et al., 2005) and retrieves observations at irregular temporal intervals. Therefore, the main limitations of conventional satellite radar and laser altimetry are low spatial resolution, local coverage (for short repeat orbit missions) and low temporal resolution (for long repeat missions such as CryoSat). In order to overcome these limitations, the forthcoming Surface Water and Ocean Topography (SWOT) satellite mission will build on the heritage of the imaging interferometric radars such as the Shuttle Radar Topography Mission (SRTM) (Kiel et al., 2006; LeFavour and Alsdorf, 2005; Rodriguez et al., 2006). However, spaceborne sensors will always face problems of: i) large ground footprints, which result in relatively low spatial resolution; ii) fixed orbit configurations, which may be inappropriate for high-resolution coverage of local water bodies; iii) coarse temporal resolution and/or the non-regular revisit intervals. These limitations restrict their ability to measure the temporal and spatial variation of the water level with the accuracy needed for determining the hydraulics of complex rivers and flood waves. Airborne LIDAR techniques have the advantages of better tracking of terrestrial water bodies,

 improved spatial resolution, clear segmentation between land and water surfaces and a higher accuracy (Schumann et al., 2008). However, airborne LIDAR surveys are expensive and their success depends on surveying conditions (e.g. topography and geometry, vegetation cover, size

 of the water body). For this reason, digital elevation models and digital surface models retrieved by airborne LIDAR are not universally available and are normally not retrieved during periods of hydrological interest such as flood events.

 UAVs (Unmanned Aerial Vehicles) and in particular micro-UAVs (payload less than 1.5-2 kg), represent the latest frontier in land and water monitoring because of low-altitude flight, low cost and flexible payload design (Anderson and Gaston, 2013). In recent years, miniaturized components (GNSS receivers, inertial measurement units, autopilots) have advanced (Watts et al., 2012), and UAVs have been used also for a wide range of hydrological applications such as fluvial monitoring; river bathymetry and photogrammetric DEM generation using very high resolution (VHR) imagery (Lejot et al., 2007); water velocity measurements using large-scale particle image velocimetry (LSPIV) (Detert and Weitbrecht, 2015; Tauro et al., 2016, 2015). Moreover, UAVs have attracted great interest for monitoring of environmental disasters and floods (Luo et al., 2015). UAVs are low-cost platforms that have unique capabilities to access hostile or inaccessible environments that need to be urgently monitored. Moreover, they ensure tracking of water surfaces better than satellite technology. However, for LIDAR and SAR systems, the tradeoff between performance, cost and size/weight is still a challenge to be solved before their application in UAV remote sensing (Colomina and Molina, 2014). In this paper, we demonstrate the possibility to acquire measurements of water level by a ranging system that includes a ranging sensor (radar, CLDS or sonar) and a GNSS receiver. The ranging technology described in this paper provides water level measurements with higher accuracy than spaceborne or airborne altimetry. Moreover, it ensures a spatial resolution ideal for measuring the two dimensional spatial variability of small rivers and their interaction with floodplains (Lee

et al., 2011). Lastly, the newly developed CLDS can acquire ranges to water surfaces when only



<span id="page-6-0"></span> The ellipsoidal height of the water surface is measured by subtracting the range measured by a ranging sensor from the vertical position retrieved by the onboard GNSS receiver. Afterwards the orthometric height can be retrieved from the ellipsoidal height if the geoid height is known (Featherstone, 2001). For the purpose of this work, a hexacopter has been assembled from TAROT-RC components and has been equipped with DJI Naza-M2 flight controller. The hexacopter is able to fly at least 12 minutes carrying a payload of at most 2 kg. The choice of the ranging sensors was constrained by: i) maximum weight of the payload, ii) a reasonable price necessary for flexible operations, iii) sensor interfaces that allow time synchronization with the GNSS receiver through a microprocessor. The selected ranging sensors included two off-the-

<span id="page-7-0"></span>

 compensation, noise tolerance and clutter rejection. Its maximum ranging capability is up to 10 m.

### *2.1.3. Camera-based laser distance sensor (CLDS)*

 This ranging sensor is a laser camera-based solution recently developed at Technical University of Denmark (Reyna Gutierrez, 2013). It weighs around 350 g. It was inspired by the measuring procedure proposed by Danko (2004). The range distance to the target is estimated by measuring the angle at which laser light enters the camera. The original methodology is expanded in this work to include corrections for tilting and rotation angles of the aircraft. An efficient automatic algorithm for identifying the laser dots on the water surface was developed. Our prototype consists of two laser pointers (100 mW laser diodes) and a complementary metal–oxide– 164 semiconductor (CMOS) camera. The camera resolution is 20.2 megapixels. The camera is triggered by the on-board single board computer (SBC) with an image rate of 1 frame every 2.5 seconds. The total manufacturing cost of this CLDS system is around 800 EUR. The current design of the distance-meter includes a digital camera mounted at the center between the two laser pointers. [Fig.](#page-8-0) *3* shows the geometrical configuration of the camera. Range to water surface is measured by illuminating the water surface with the laser pointers and taking a picture of the illuminated water surface. When light emitted by laser pointers hits the water surface, bright dots are formed at the interface between water and air. Due to scattering processes (in particular Rayleigh and Mie scattering), some portion of the radiation is reflected in the direction of the camera and an estimation of the range to water surface is possible.

<span id="page-8-0"></span>

**Fig. 3.** 

176 The angle  $\alpha$  is a design parameter. The CLDS was built with  $\alpha = 90^\circ$  to simplify the measuring

177 concept and the derivation of the formulas. The CLDS shown in [Fig.](#page-8-0) *3* is exactly symmetrical.

178 Indeed, only one laser would be sufficient to acquire the range to the surface; nevertheless, two

179 laser pointers improve error assessment and system accuracy.

180 The value of the measured range Hm can be computed by measuring the angle θ′ at which light

181 enters the camera, i.e. from equation [\(1\).](#page-9-0)

<span id="page-9-0"></span>
$$
Hm = \frac{A}{\tan \theta'}\tag{1}
$$

182

183 Alternatively, the measured range Hm can be obtained through equation [\(2\)](#page-9-1)

<span id="page-9-1"></span>
$$
Hm = \frac{A \cdot f}{ImD} \tag{2}
$$

184 Where ImD (Image distance) is the distance between the center of the image and the recorded

185 light source. A calibration procedure is needed to convert from the number of pixels from the

186 center of the image (PFC) to ImD as shown in equation [\(3\)](#page-9-2)

<span id="page-9-2"></span>
$$
ImD = PFC \cdot d_{pp1} + d_0 \tag{3}
$$

187 Where  $d_{\text{ppl}}$  and  $d_0$  are the coefficients of the first-order polynomial producing the best least-

188 squares fit to the data. Equations (2) and [\(3\)](#page-9-2) can be applied only when the focal length (f) of the

189 camera is exactly and the focus is constantly set to infinity. Otherwise, the calibration procedure

190 needs to estimate the angle θ′ directly from the number of pixels (PFC) as shown in equation [\(4\).](#page-9-3)

<span id="page-9-3"></span>
$$
\theta' = PFC \cdot r_{pp1} + r_0 \tag{4}
$$

191 Where  $r_{\text{ppl}}$  and  $r_0$  are the coefficients of the first-order polynomial producing the best least-192 squares fit to the data. The calibration procedure, which has to be performed to estimate the  $r_{\text{ppl}}$ 193 and  $r_0$  coefficients, is presented in the appendix. The calibration procedure allows estimation of the angle θ′ by measuring PFC, without having to consider the linear or nonlinear intrinsic camera parameters, such as focal length and lens distortion.

Onboard the UAV, tilting and rotation cause a displacement of the light sources from their

equilibrium position. The changes in the geometrical relationships generate an error in the

estimation of the true range distance (hereafter defined as Ht) between the sensor of the camera

and the water surface. Tilting is the angle between the plane on which the camera and laser are

located, i.e. the axis of the CLDS, and the horizontal plane (angle β as shown in [Fig](#page-10-0)*. 4*). Rotation

201 occurs between the vertical line and the optical axis of the camera (angle  $\delta$  as shown in [Fig. 5\)](#page-10-1).

<span id="page-10-0"></span>

<span id="page-10-1"></span>Fig. 4 *Fig. 5*

 If tilting pushes the light source below the axis of the distance meter, formula [\(5\)](#page-11-0) can be used to obtain the true range (Ht) between the camera and the water surface:

$$
Ht = [(Hm + A \cdot \tan \beta) \cos \beta] \cdot \cos \delta
$$

<span id="page-11-0"></span>*(5)*

<span id="page-11-1"></span>*(6)*

 Conversely, if the tilting pushes the light source above the axis of the CLDS, formula [\(6\)](#page-11-1) can be used:

$$
Ht = [(Hm - A \cdot \tan \beta) \cos \beta] \cdot \cos \delta
$$
 (0)

 If pitch and roll angles are retrieved on board the UAV, the measured range can be corrected according to equation [\(5\)](#page-11-0) and [\(6\)](#page-11-1) (Reyna Gutierrez, 2013). If the angles are not retrieved on board, the resulting error on the range can be estimated as shown in [Fig. 6.](#page-11-2) Numerous tests have been conducted in order to determine the best configuration of the CLDS in terms of: i) arm length A, ii) wavelengths of the two laser pointers, iii) optimal camera configuration parameters 211 such as optical zoom and resolution.

- <span id="page-11-2"></span> The arm length choice affects the measuring range function, as shown in [Fig. 7.](#page-11-3)
	- **Fig. 6**

<span id="page-11-4"></span><span id="page-11-3"></span>
$$
Fig. 7
$$

- [Fig.](#page-11-3) **7** shows that the resolution of the measurements depends on the derivative of the range
- function. Hence, a longer arm will result in higher resolution, especially for longer ranges.
- Indeed, in [Fig.](#page-11-3) **7**, the smoothest curve is for an arm length of 0.6 m. However, the payload size of
- small UAVs is limited and thus a 30 cm arm was chosen for our tests. The wavelengths of the
- two laser pointers were chosen as 450 nm and 531 nm, because reflectivity of water is relatively

 high at these wavelengths as a consequence of the optical proprieties of water as described in Hale and Querry (1973).

 When the laser light hits the water surface, a bright dot is formed at the point of contact. However, additional bright spots might be visible due to reflection from the riverbed and due to 222 additional scattering processes caused by water waves. To identify the two dots formed by laser reflection, an automatic identification algorithm was developed consisting of the following computational steps: i) the RGB image is converted to Hue, Saturation and Value (HSV) image. Quasi-circular shapes in the image are found through circular Hough transform (Yuen et al., 1990). In case there are multiple circles in the image, the two circles (one generated by the left laser and one by the right laser) with the highest mean Value (V) are considered to be the contact spots. Thereafter, ii) the brightest pixel (pixel with the highest Value) is identified inside each of the two circles (laser dots). The brightest pixel typically lies in the center of the laser dot in case of normal light incidence. Lastly, iii) the distance (PFC) between the center of the image and the two identified brightest pixels is computed. Post-processing of the images is performed after the flight and takes around 30 seconds per image.

#### *2.1.4. GNSS system*

 The differential GNSS system consists of two NovAtel receivers: one used as master station (flexpack6) and one as rover (OEM628 board). A NovAtel GPS-703-GGG pinwheel triple frequency and GLONASS antenna is used as base station and an antcom (3G0XX16A4-XT-1-4- Cert) dual frequency GPS and GLONASS flight antenna is used as rover station on the UAV. Raw pseudoranges and carrier phase measurements are stored at 5 Hz. The position solution is post-processed using Leica Geomatic Office v 8.1 in kinematic mode. In post-processed mode, a Kalman filter can be applied both in forward and backward direction for best position

 performance. The length of the GPS baseline affects the vertical and horizontal accuracy of the drone position. Position error is expected to increase by 1-3 ppm (1-3 mm additional error per km of baseline).

*2.1.5. Payload controller*

 Data acquired by the different sensors are saved on the SBC (BeagleBone Black) and a time synchronization of the different sensors can be performed. Synchronization between the position retrieved by the GNSS system and the range retrieved by the sensors is essential for accurate water level observations, as described in Appendix B.

#### *2.2. Testing of the sensors*

 To test the accuracy of the system, both static (ground-based) and dynamic (airborne) tests were performed. First, several tests were conducted from bridges of different heights over free-flowing rivers in order to test accuracy, precision and maximum ranging capability. Beam divergence was tested by acquiring measurements inside a water well of small diameter. After the ground- based tests, numerous flight tests were conducted over a lake. Because the water level in the lake can be assumed to be uniform in space, these flights allowed determination of the accuracy of the full system, which consists of the GNSS receiver and the ranging sensors. Appendix B reports the experimental settings of both static and airborne tests.

#### 261 *2.2.1. Ground-based evaluation*

262 Accuracy of the ranging sensors was estimated using as reference a water level dip meter, which 263 has an accuracy better than 0.3% of the range. When tested in static mode, sensors acquired 264 measurements for 30 seconds. Subsequently the average range  $(\bar{x})$  was computed as the weighted 265 arithmetic mean as shown in equation [\(7\)](#page-14-0) after outlier removal ( $\geq$ 5 $\sigma$ ).

<span id="page-14-0"></span>
$$
\bar{x} = \frac{\sum_{i=1}^{N} f_i x_i}{\sum_{i=1}^{N} f_i}
$$
\n<sup>(7)</sup>

266

267 In equation [\(7\)](#page-14-0)  $x_i$  is an observation and  $f_i$  the frequency of that value. N is the total number of 268 measurements which depends on the reading range of the individual sensor.

269 Precision is estimated as standard deviation  $\sigma$  of the measuring stack, and is computed using 270 equation [\(8\):](#page-14-1)

<span id="page-14-1"></span>
$$
\sigma = \sqrt{\frac{\sum_{i=1}^{N} f_i \cdot (x_i - \bar{x})^2}{\sum_{i=1}^{N} f_i - 1}}
$$
\n(8)

271

272 Maximum ranging capability is the maximum range from which the sensor can retrieve a

273 measurement with a reasonable accuracy (i.e. 5% of the range).

274 Beam divergence is defined as the measure (in angular units) of the increment in [beam](https://en.wikipedia.org/wiki/Beam_diameter) 

275 [diameter](https://en.wikipedia.org/wiki/Beam_diameter) with distance from the [optical aperture](https://en.wikipedia.org/wiki/Aperture) or antenna from which the sonic or

276 electromagnetic beam emerges. A larger beam divergence leads to a larger ground footprint of

277 the signal, which results in contamination of the signal if the surface is inhomogeneous. For the

- 278 CLDS this parameter is negligible, since its ground footprint directly depends on the arm length
- 279 A and the laser beam divergence is very low. Moreover, the CLDS provides images of each
- 280 individual acquisition and the user can perform a-posteriori supervision to control if the

 measured target is indeed the water surface. For the radar and the sonar, beam divergence is a critical parameter to ensure that water is measured without interference from the surroundings. This parameter has to be considered in order to monitor water bodies (e.g. large sinkholes, rivers surrounded by dense vegetation), which only expose a narrow stretch of water to aerial view. Indeed, because of loss of GNSS signal, flights under vegetation canopy or inside small cavities (e.g. karst sinkholes) cannot be performed without losing position accuracy. Beam divergence was estimated by acquiring measurements over water wells of small diameter, while water was gradually being pumped out, as described in Appendix B.

# *2.2.2. Airborne evaluation*

290 Numerous flights were conducted above a  $0.02 \text{ km}^2$  lake located near Holte, Denmark (55.821720°N, 12.509067°E). Water level in the lake is practically uniform. Whilst the sonar and the CLDS identify only one target in the field of view, the radar can identify multiple targets and reports the target angle for each of those. This requires an accurate identification of the target, which is representative of the water surface. Indeed, sometimes multiple targets are retrieved at nadir angle, for instance when vegetation is overhanging the water body. In that case, postprocessing requires switching between different targets to obtain a result that is continuous in time. Moreover, a low-pass digital filter was applied on the 15Hz raw radar data. A weighted moving average (WMA) with a temporal window of 0.33 s (five observations) was applied to smoothen the signal as shown in equation [\(9\).](#page-15-0)

<span id="page-15-0"></span>
$$
WMA_t = w_1 A_{t-2} + w_2 A_{t-1} + w_3 A_t + w_4 A_{t+1} + w_5 A_{t+2}
$$
 (9)

Weights (w1, w2…w5) are normally set to a high value for the measurement taken at the actual

time (At) and to lower values for the previous and subsequent measurements.

.The overall accuracy of the system consisting of the GNSS receiver and the ranging sensor

- ( $\sigma_{tot}$ ) is assumed to be that of two independent normally distributed variables: the ranging sensor
- accuracy and the GNSS accuracy [\(10\).](#page-16-0)

<span id="page-16-0"></span>
$$
\sigma_{tot} = \sqrt{\sigma_s^2 + \sigma_{RTK}^2} \tag{10}
$$

307 where  $\sigma_s$  is the accuracy of the ranging sensor and  $\sigma_{RTK}$  is the accuracy of the GNSS receiver.

# **3. Results**

 The first section of the results describes the technical performance of the ranging sensors when tested from a static position on the ground. Results are based on numerous tests conducted from bridges of different heights to compare the technical performance of the different sensors. The second section describes the results of the flight tests that are intended to evaluate the accuracy of the integrated system, i.e. GNSS receiver and sensors operating on board the UAV.

#### *3.1. Ground-based performance results*

 Sensors demonstrated different performance in terms of accuracy and standard deviation of the measuring stack when tested from bridges of different heights. Appendix B lists the experimental settings for the static tests. [Fig. 8](#page-17-0) shows that the sonar usually tends to overestimate the range to water surface, which is probably caused by a slight penetration of the ultrasonic wave (42 kHz) below the water surface. Conversely, the radar usually tends to underestimate the range. The

<span id="page-17-1"></span><span id="page-17-0"></span>

<span id="page-18-0"></span>Fig.9

 Finally, the accuracy of the retrieved vertical position has to be assessed. The accuracy of the GNSS height depends mainly on: i) the integer ambiguity solution that has to be fixed to obtain reliable observations, ii) the satellite geometry that affects the dilution of precision (DOP), iii) multipath interference, especially because of signal reflection from the water surface. *3.2. Airborne performance results* In this section, we report the observations of two flights and we show a table summarizing the entire dataset of flights over the lake. The range measured by each of the sensors and the altitude retrieved by the GNSS are shown in [Fig.10.](#page-18-1) The figure contains the entire dataset of observations retrieved by the radar and sonar. Only not-a-number (NaN) values are removed. The sonar outputs NaN when the range exceeds the maximum range capability (10 m). For the CLDS, we only reported the measurements retrieved from images in which the laser dots are clearly identifiable on the water surface. Fig.10 [Fig.10](#page-18-1) shows an extremely high correlation (Pearson coefficient of 0.9991), between the GNSS

<span id="page-18-1"></span> and the radar measurements, which indicates the consistency of our ranging technology. The laser dots are generally distinguishable on the water surface only when the range to water surface is less than 12-13 m . Similarly, the sonar provided accurate measurements only when the UAV

 was hovering at low altitudes (less than 10 m from the water surface). Indeed, the radar and sonar curves only overlap during these flight maneuvers.

367 In [Fig. 11](#page-19-0) we display the water level measured by the different sensors. Outliers ( $>2\sigma$ ) were removed.

<span id="page-19-0"></span>Fig. 11

 Mode value, mean and standard deviation of water level retrieved by each of the sensors are reported in [Table](#page-21-0) 2 under the column with flight date "04/04". The dispersion in water level measurements retrieved by the system consisting of the radar and the GNSS receiver may be due to multipath errors on the GNSS receiver. The cut-off angle for the elevation of the satellites, which defines the angle below which GPS satellites are excluded, turned out to be a sensitive parameter. The selected values for each flight are reported in Appendix B. The water level values retrieved by the sonar had low accuracy, especially during high-speed maneuvers. Since the range to water surface was greater than the maximum range capability of the sonar for a significant portion of flight duration, the sonar retrieved many NaN values and noisy observations. However, the mode value retrieved by the sonar is 24.14 m, which is close to the mean value retrieved by the radar. The CLDS exhibits only few observations due to limited range capability and low frame rate. Moreover, natural light conditions complicate the recognition of the laser dots on the water surface. In order to estimate the absolute accuracy of the sensors, results were compared to in-situ

measurements of water level. For the in-situ measurement, an additional accurate RTK (Real

 Time Kinematic) GNSS rover station was used, which was connected to a Danish GPS network. The position was averaged over a period of one minute which resulted in 24.10 m above the DVR90 geoid model (with an estimated accuracy of the GNSS rover station of around 5-6 cm). For this flight, the accuracy of the radar is thus better than 5 cm, the mode value of the sonar is around 4 cm from the ground truth, while the mean value retrieved by the CLDS is within two decimeters.

 The second flight reported in [Fig. 12](#page-20-0) evaluated performance for higher drone altitude (up to 60 m) above the water surface.

<span id="page-20-0"></span>Fig. 12

 As shown in [Fig. 12,](#page-20-0) the radar and the GNSS show very high correlation for the entire flight. The flight confirmed the limited ranging capability of the sonar (specified as 10 m, but already very noisy beyond 9 m). The CLDS retrieved ranges up to 13 m, however standard deviations increased significantly with range. In [Fig. 13](#page-20-1) we compare the water level retrieved by the three different sensors for this flight.

<span id="page-20-1"></span>Fig. 13

 Statistics of the flight are shown in [Table](#page-21-0) 2 under the column "27/05". In-situ water level was 24.01 m. [Fig. 13](#page-20-1) shows that the sonar measurements were unsuccessful. The CLDS, despite very high standard deviations, shows a mean value that is very close to the ground truth. The radar shows higher dispersion for long ranges. Moreover, systematic error is still observable, in fact

<span id="page-21-0"></span>



<span id="page-22-0"></span>

 Jason-2 (Asadzadeh Jarihani et al., 2013), Envisat (Frappart et al., 2006) and Cryosat-2 (Song et al., 2015) is in the order of some tens of dm. Moreover, satellite radar altimetry generally has a spatial resolution lower than satellite laser altimetry and requires that rivers are hundreds of meters wide to avoid signal contamination by interfering land and vegetation (Maillard et al., 2015). With UAV-borne monitoring, water surface and interfering surroundings can be clearly separated due to the smaller ground footprint, and the possibility to retrieve individual radar target angles. However, for very narrow fields of view, the CLDS is the only sensor that can provide reliable water level measurements. Image analysis as part of the post-processing workflow ensures that measurement are accepted only if the monitored target is the water surface. This is the case for rivers surrounded by dense riparian vegetation or for small targets such as karst sinkholes, e.g. on the Yucatán Peninsula (Gondwe et al., 2010). Our CLDS solution overcomes the limitations of traditional red wavelength time-of-flight (TOF) laser distance meters, which are not suitable for ranging to water surfaces, because the reflectivity of water is very low for red visible wavelengths.

 Only ground-based hydrometric stations ensure an accuracy higher than the one achieved with UAV-based monitoring, but coverage and reliability of in-situ monitoring networks have been degrading in many regions of the world. Moreover, despite providing high accuracy and temporal resolution, in-situ stations acquire only local measurements and tend to fail during extreme events. Therefore, UAV-based water level monitoring is beneficial for the monitoring of a wide range of hydrological systems, including small-scale rivers, ephemeral lakes, sinkholes, meltwater lakes, etc… UAV-based water level observations can resolve the spatial multidimensional variability of rivers. Indeed, UAVs can monitor water level along and across the river course, in order to obtain water slope and assess interaction between rivers and adjacent

 floodplains. Improved sharpness and reliability of estimates of surface water-groundwater interaction using UAV-based monitoring of river water levels have already been reported (Bandini et al., 2016). Furthermore, UAVs can sense water level in unconventional remote sensing targets such as sinkholes or cenotes. This could potentially improve mapping of phreatic surfaces, for instance for the Yucatan peninsula (Bauer-Gottwein et al., 2011). Additionally, UAVs can potentially be used during extreme events when in-situ monitoring stations often fail and satellite observations do not ensure the required spatial and temporal resolution. Thus, UAVs 485 have the potential to improve flood risk assessment. However, the  $\pm$ 7cm accuracy of our technology may still be insufficient for rivers flowing through low-lying terrain. Nonetheless, the accuracy is better than other spaceborne and airborne technologies and UAVs have a great potential in improving flood mapping because they allow optimal timing of the observations and high spatial resolution. UAV-based observations of water level in the flooded areas allow determination of stage-damage curves (Cammerer et al., 2013) which are essential for the design of insurance policies.

### **5. Conclusions**

 UAV-based remote sensing of river and lake water level (orthometric height) has the potential to fill the gap between in-situ measurements and spaceborne remote sensing. It ensures: i) high accuracy, ii) optimal spatial resolution, iii) flexible timing of the sampling, and iv) precise tracking of lakes and rivers. Different water surface ranging sensors were tested: a radar, a sonar, and a CLDS.

Static (on ground) and dynamic (airborne) tests demonstrated the following results:



- The master thesis from Reyna Gutierrez, J. A. (2013) ''Monitoring and modeling of regional
- groundwater flow on the Yucatán Peninsula'' can be obtained from the authors upon request.

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# **Appendix A. Calibration of the CLDS**

- The CLDS needs to be calibrated in order to provide a ranging measurement. Calibration has
- been performed acquiring multiple range measurements (from 0 to 12 m) using a black vertical
- wall as calibration target. Since the focal length of the camera is not exactly known, equation [\(4\)](#page-9-3)
- 648 must be used and the calibration is used to retrieve the coefficients  $r_{\text{pp1}}$  and  $r_0$  for converting from
- 649 pixel units to angular units. The relationships between  $\theta'$  and the distance from the laser dots to
- the center of the image (PFC) are shown in [Fig. A.1](#page-30-0) for each of the laser pointers. Alternatively,
- [Fig. A.2](#page-30-1) depicts the relationship between the range to the target and PFC.

<span id="page-30-1"></span><span id="page-30-0"></span>

 [Fig. A.1](#page-30-0) and [Fig. A.2](#page-30-1) show that the laser pointers' curves are not coincident as a consequence of the slight asymmetry of the layout (imaging sensor of the camera not placed exactly in the middle of the two laser pointers). As confirmed by [Fig. A.1,](#page-30-0) the relationship between PFC and the measured angle is approximately linear for each of the two laser pointers. Calibration has shown an r (Pearson linear correlation coefficient of determination) of 0.99978 and an RMSE (Root Mean Square Error) of 7.16 cm for the blue laser (left laser); an r of 0.99937 and an RMSE of 8.29 cm for the green laser (right laser). Calibration error is displayed in [Fig. A.3.](#page-30-2) Fig. A.3 Fig. A.4 

<span id="page-30-3"></span><span id="page-30-2"></span>[Fig. A.](#page-30-2)*3* demonstrates that the advantage of using two laser pointers is improved error

assessment. Considering the average of the measurements of the two laser pointers, calibration

 RMSE is reduced to 5.61 cm. When range to water surface has to be retrieved, the precise computation of PFC is more problematic than during the simple calibration procedure. Indeed, while laser dots can be normally identified as in [Fig. A.4](#page-30-3) (a), laser dots on the water surface might have contours that are less defined as in [Fig. A.4](#page-30-3) (b). Sometimes even multiple laser dots are visible, as shown in [Fig. A.4](#page-30-3) (c). This is caused by: i) atmospheric scattering processes, ii) scattering processes due to water waviness iii) vibrations of the UAV. The laser light reflected from the bottom is occasionally visible in the image, especially in case of shallow or very clear water, as shown in [Fig. A.4](#page-30-3) (d). Experiments showed that the uncertainty in the PFC increases 674 with the range to water surface. This is displayed in [Fig.](#page-32-0) A.5 with the curve PFC- $\sigma_{PFC}$ . Fig. A.5 clearly shows that the green laser exhibits larger uncertainty than the blue laser since green wavelengths are scattered to a greater extent than blue wavelengths. The expected uncertainty in the range can be estimated using the derivative of the range function as shown in equation [\(A.1\)](#page-31-0).

<span id="page-31-0"></span>
$$
\sigma(range) = \frac{\partial range}{\partial (PFC)} \sigma(PFC) \tag{A.1}
$$

<span id="page-32-1"></span>

<span id="page-32-0"></span>

 In [Table](#page-33-0) B.1 we report the location, the date and time of the day, the environmental conditions and the water flow speed for each of the static tests. The mean value and the standard deviation of the measurements are shown in Figure 8.

<span id="page-33-0"></span>Here Table B.1

 Illumination conditions are reported in the table because they affect visibility of the laser dots on the water surface. This factor has been critical only in case of sun glint conditions during which laser dots are hardly identifiable. On the other hand, wind stress and current can affect water surface roughness and change the intensity of the backscattered radar signal.

 Estimates of beam divergence for the different sensors were obtained from tests above a cylindrical water well of diameter (D) equal to 0.7 m. The sensors were placed exactly in the middle of the water well as shown in Fig. [B.1.](#page-33-1) The initial range between the sensors and the water surface was 0.5 m. Subsequently, the well was pumped to gradually increase the range to 701 the water surface. Beam divergence  $(\varphi)$  was then computed according to equation B.1.

$$
\varphi = 2 \cdot \tan^{-1} \frac{D}{2 \cdot r_c} \tag{B.1}
$$

702 In equation B.1,  $r_c$  is the critical range i.e. the range at which the sensor first produced erroneous results because of interference with the well walls. [Fig.](#page-33-1) B.1 provides an illustration of the experimental setup.

<span id="page-33-1"></span>Fig. B.1

<span id="page-34-0"></span>





739 Fig. 1. Illustration of measurement principle for retrieving water level. The system includes: i) the UAV, ii) the 740 sensors to measure the range from the UAV to the water surface, iii) a GNSS receiver on board the U 740 sensors to measure the range from the UAV to the water surface, iii) a GNSS receiver on board the UAV providing<br>741 ccurate vertical and horizontal position. Centimeter-level position accuracy is obtained through the i 741 accurate vertical and horizontal position. Centimeter-level position accuracy is obtained through the installation of an in-situ GNSS master station providing corrections for a kinematic post-processed solution. an in-situ GNSS master station providing corrections for a kinematic post-processed solution.



 Fig. 2. Picture of the drone payload. It includes the three tested sensors (CLDS, radar and sonar), the GNSS system (antenna and receiver), the IMU, the Single Board Computer (SBC) and the power convertion units (DC/DC converters).



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 Fig. 3. Geometric configuration of the CLDS solution. A is the distance between the center of the camera and each 754 of the laser pointers.  $\alpha$  is the angle between each of the lasers and the focal plane of the camera. Hm is the distance between the center of the image focal plane and each  $\overline{755}$ between the camera and the water surface. ImD is the distance between the center of the image focal plane and each of the recorded laser light dots. f is the focal length of the camera. θ′ is the reflection angle. θ is its angle between the 757 axis of the CLDS and the reflected ray. γ is the angle between incident and reflected ray. If  $\alpha$  is 90° (as in the figure), γ is equal to θ'. figure),  $\gamma$  is equal to  $\theta'$ .







 

787 Fig. 8. Absolute error as a function of the range measured by each of the ranging sensors. Absolute error is computed using the water level dip meter as reference. The marker is the average error (bias) of all measurem 788 computed using the water level dip meter as reference. The marker is the average error (bias) of all measurements taken for a specific range, while the bar shows the standard deviation. taken for a specific range, while the bar shows the standard deviation.





 Fig. 9. Sonar and radar errors as a function of the range. Dots represent the measurements acquired by the radar and the sonar. The regression line shows that the absolute error is a function of the range.

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801 Fig. 10. Observations retrieved during the flight on April 4, 2016. The plot shows the range measured by the radar (blue), sonar (red), CLDS (green) in meter (m) to the water surface, and the drone altitude retrieved b 802 (blue), sonar (red), CLDS (green) in meter (m) to the water surface, and the drone altitude retrieved by the GNSS (black) in meter above mean sea level (mamsl). (black) in meter above mean sea level (mamsl).



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807 Fig. 11. Water level (mamsl) observations retrieved during the flight on April 4, 2016. Each of plots shows the water<br>808 level observations measured by subtracting the range retrieved by each of the sensors (radar, so 808 level observations measured by subtracting the range retrieved by each of the sensors (radar, sonar, CLDS) from the 809 GNSS altitude. In each plot, the black line is the mean of the water level observations and the ma GNSS altitude. In each plot, the black line is the mean of the water level observations and the magenta line is the 810 mode of those observations.



 Fig. 12. Observations retrieved during the flight on May 27, 2016. The plot shows the range measured by the radar (blue), sonar (red), CLDS (green) in meter (m) to the water surface, and the drone altitude retrieved by the GNSS (black) in meter above mean sea level (mamsl).



819 Fig. 13. Water level (mamsl) observations retrieved during the flight on May 27, 2016. Each of plots shows the water level observations measured by subtracting the range retrieved by each of the sensors (radar, sonar, water level observations measured by subtracting the range retrieved by each of the sensors (radar, sonar, CLDS) 821 from the GNSS-derived altitude. In each plot, the black line is the mean of the water level observations and the 822 magenta line is the mode of those observations.

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 Fig. A.3. Calibration error for left laser (blue column), right laser (green column) and for the average (red column) between the two laser pointers.

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Fig. A.4. Airborne image of water surface taken by the CLDS. (a) the two laser dots are clearly identifiable (b) larger laser dots with contours that are less identifiable (c) multiple green laser dots caused by multiple r larger laser dots with contours that are less identifiable (c) multiple green laser dots caused by multiple reflection and scattering processes (d) laser light is reflected by the bottom (larger dots) and by the surface (smaller dots)





 $\frac{850}{851}$ Fig. A.5. Uncertainty ( $\sigma_{PFC}$ ) in computing the number of pixels as a function of PFC, for green and blue laser.







865 Fig. B.1. Schematic representation of the test conducted over the water well to retrieve beam divergence ( $\varphi$ ) for each of the sensors. D is the diameter of the water well,  $r_c$  is the critical range. of the sensors. D is the diameter of the water well,  $r_c$  is the critical range.

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# <sup>868</sup> **Tables**

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871 *Table 1. Technical performance of the sensors and of the GNSS receiver when tested in static mode.*

	mean absolute	standard	<b>Maximum</b> ranging	<b>Beam divergence</b>	
	error (percentage	deviation of the	distance		
	of the range)	stack			
Radar	$-1.09\%$	$0.064 \; \mathrm{m}$	60 m near field	$\sim 30^{\circ}$	
			200 m far field		
<b>Sonar</b>	0.98%	$0.007 \text{ m}$	10 <sub>m</sub>	$>40^{\circ}$	
<b>CLDS</b>	1.5%	2.3 % of the range	13 <sub>m</sub>	negligible	
<b>GNSS</b> receiver	negligible	Vertical			

#### coordinates : 4-6

#### cm at 2 sigma



- 873
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- 875 *Table 2. Summary of the test flights over the lake. Each flight is named with the date (corresponding year is 2016). Ground truth*
- 876 *was measured with a RTK GNSS rover station connected to the network of reference stations. Statistics concern the water level*
- 877 *observations measured by subtracting the GNSS flight altitude from the range to water surface measured by each of the sensors.*
- 878 *Statistics are computed after removal of the observations that lie beyond 2σ.*

#### **Flight date (dd/mm/2016)**





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## 891 *Table 3. Accuracy and ground footprint of different techniques for observing water level*



#### 892

893 *Table B.1. Locations, settings and environmental conditions during static (on ground) tests. Coordinates are in WGS84. Country* 

894 *is either Denmark (DK) or Italy (IT). Range (m) is the value measured by the water level dip meter. Water speed has qualitatively* 

895 *been classified into no speed (still water), low (less than 0.4 m/s), medium (between 0.4 and 1 m/s), and high speed (more than 1 896 m/s). Wind speed has been qualitatively classified into no wind, low (wind less* 

896 *m/s*). Wind speed has been qualitatively classified into no wind, low (wind less than 2 m/s), medium (between 2 m/s and 8 m/s), <br>897 and high wind speed (more than 8 m/s). Illumination has been qualitatively classifie 897 and high wind speed (more than 8 m/s). Illumination has been qualitatively classified into artificial lightening, low (less than 20 **898** 000 lux), medium (between 20 000 lux), and high illumination (more than 50 000 l 898 *000 lux), medium (between 20 000 and 50 000 lux), and high illumination (more than 50 000 lux)*



		Parmigiana,							
45.029723	10.959166	Canale della <b>Bonifica</b> Reggiana Montovana	<b>IT</b>	7.10	22/12	14:05	low	low	low
45.029726	10.960432	Canale della <b>Bonifica</b> Parmigiana	IT	7.33	22/12	9:30	low	low	low
44.650573	10.794755	Secchia	IT	9.79	29/10	12:00	medium	medium	medium
44.821261	10.994579	Secchia	IT	11.16	29/10	12:50	medium	medium	high
44.67578	10.860146	Secchia	IT	12.20	29/10	13:50	medium	medium	medium
45.008365	10.977453	Secchia	IT	12.72	29/10	20:30	medium	medium	low
44.727259	11.045292	Panaro	IT	12.97	29/10	8:30	medium	low	low

<sup>899</sup>

# 900 Table B.2. Summary of the test flights over the lake.



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